Basic Approaches in Assessing Combustion Stability to Pressure High-frequency Oscillations in Liquid Rocket Engines

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The paper presents main principles of the procedure for assessing com stability to high-frequency (acoustic) pressure oscillations in liquid propellant and gas generators. Stability margin assessment is based on experimentally stability characteristics and their comparison with maximum allowable Small-amplitude decrement determined by natural acoustic noise of a c process as well as relaxation time of combustion response to artificial pressu disturbances are used as the stability characteristics.